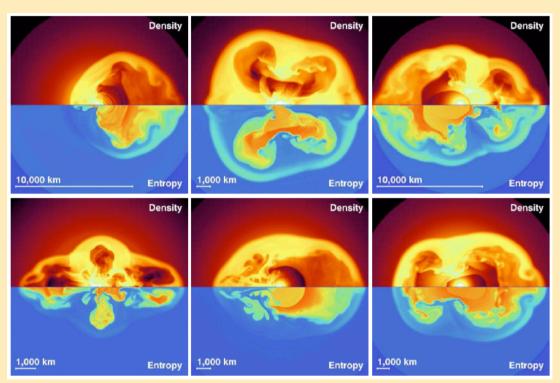
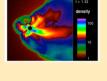
## The basics of the Advective-Acoustic Cycle









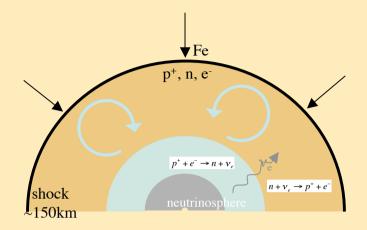
Chanaud & Powell (1965)

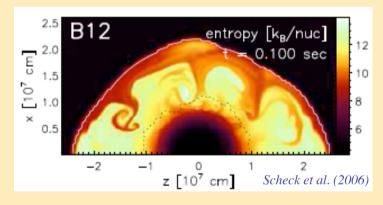
*Janka et al.* (2004)

Thierry Foglizzo, Pascal Galletti (CEA Saclay) Thomas Janka, Leonhard Scheck (MPA Garching)

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## Instabilities during the phase of stalled accretion shock

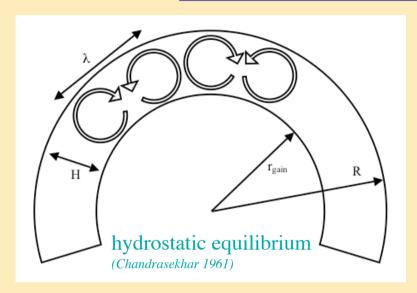


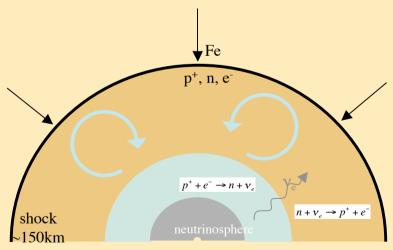


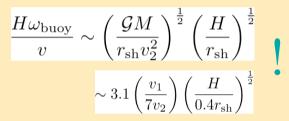
- Convection in the gain region, low 1 (Herant, Benz & Colgate 1992)
- l=1,2 SASI in an adiabatic flow: vortical-acoustic cycle (Blondin, Mezzacappa & DeMarino 2003) or purely acoustic mechanism (Blondin & Mezzacappa 2006)?
- -Neutron star kick resulting from the l=1 instability of convection and/or vortical-acoustic cycle
  -(Scheck et al. 2004, Janka et al. 2004, Scheck et al. 2006)
- New explosion mechanism driven by acoustic waves, initiated by the advective-acoustic cycle (*Burrows et al. 2005*)

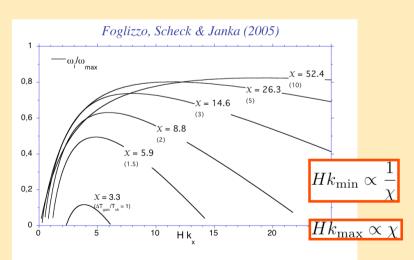
convection?
SASI 2003?
SASI 2006?
advective-acoustic cycle?

## Contribution of the convective instability to a mode l=1?





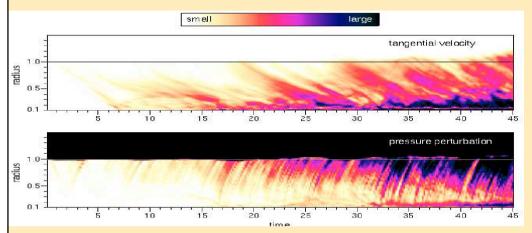




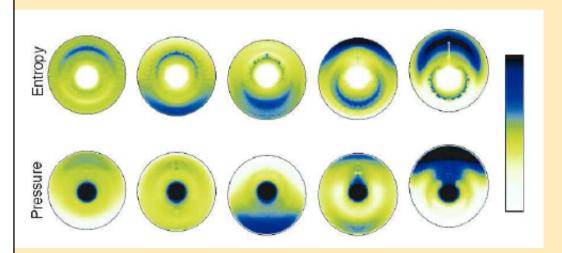
$$\chi \equiv \int_{z_{\rm sh}}^{z_{\rm gain}} \omega_{\rm buoy} \frac{\mathrm{d}z}{v}$$



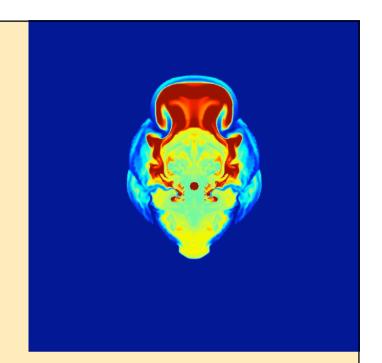
## Asymmetry without convection: adiabatic simulation of a stalled accretion shock on a neutron star



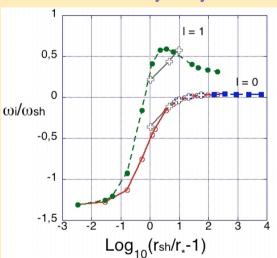
Evidence for a vortical-acoustic cycle (Blondin, Mezzacappa & DeMarino 2003)



A purely acoustic cycle ? (Blondin & Mezzacappa 2005)



#### Linear stability analysis



1=0: Houck & Chevalier (1992) 1=0,1: Galletti (PhD Thesis 2005) Blondin & Mezzacappa (2005)

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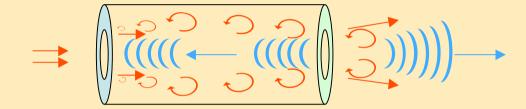
vibrations in Ariane 5: segmented solid propergol Mettenleiter, Haile & Candel (2000) J. of Sound and Vibration 230, 761

impinging shear layers

Rockwell, D. 1983, AIAA J., 21, 645

#### Aero-acoustic instabilities

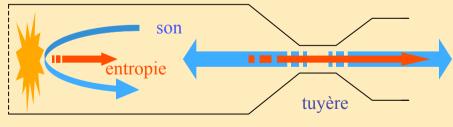
- advected perturbations
- acoustic feedback
- vortical-acoustic cycle





whistling kettle Chanaud & Powell (1965) J. Acoust.Soc. Am. 37, 902

#### • entropic-acoustic cycle



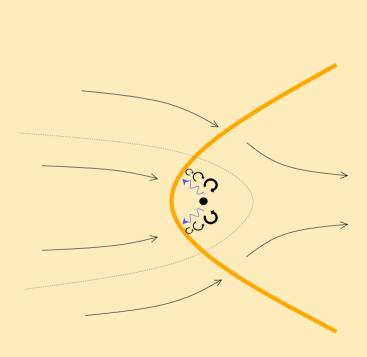
combustion

#### rumble instability of ramjet combustors

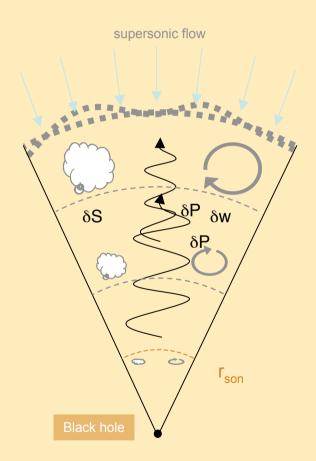
Abouseif, Keklak & Toong (1984) Combustion Science and Technology, 36, 83

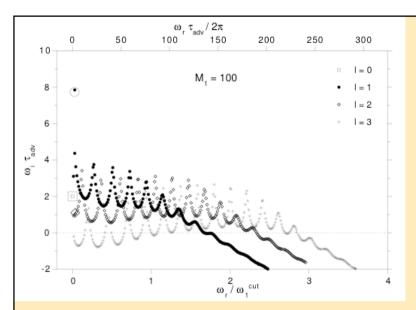
## Analytical study of the advective-acoustic coupling in a radial, accelerated flow

Foglizzo (2001, 2002), Foglizzo, Galletti & Ruffert (2005)



Bondi-Hoyle-Lyttleton accretion; unstable





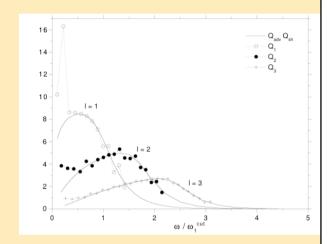
#### isothermal flow (Foglizzo 2002)

# $Q_{\rm sh} \qquad \delta S \qquad \delta p^{-}$ $R_{\rm sh} \qquad \delta p^{+} \qquad R_{\nabla} \qquad \delta p^{-}$

## $\mathcal{Q}e^{i\omega\tau_{\mathcal{Q}}} + \mathcal{R}e^{i\omega\tau_{\mathcal{R}}} = 1$

#### Beyond the eigenspectrum

- identification of the 2 cycles  $(Q,\tau_Q)$  and  $(R,\tau_R)$ 



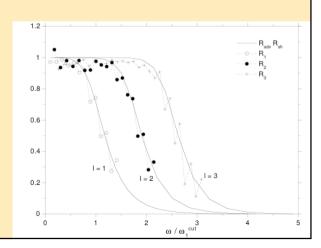
#### purely acoustic cycle

advective-acoustic cycle

efficiency  $\mathcal{Q} \equiv \mathcal{Q}_{
m sh} \mathcal{Q}_{
abla}$ 

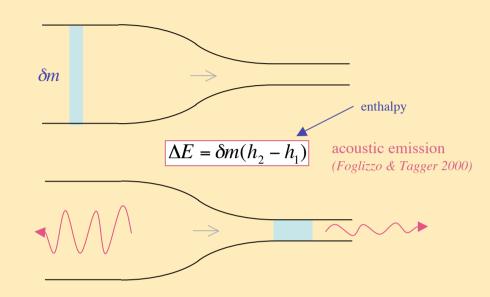
duration  $au_{Q}$ 

efficiency  $\mathcal{R} \equiv \mathcal{R}_{\mathrm{sh}} \mathcal{R}_{
abla}$  duration  $\mathcal{T}_{\mathcal{R}}$ 



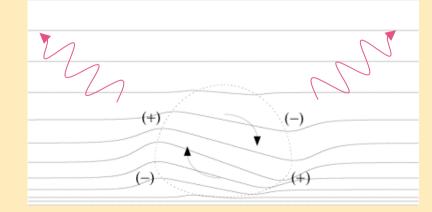
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#### Advective-acoustic coupling: hand waving

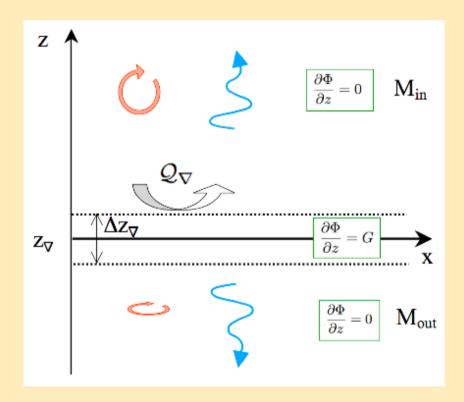


advection of entropy

advection of vorticity



#### Advective-acoustic coupling: illustrative simulations



2 isothermal simulations (thanks to F. Masset)

-> accelerated flow  $(M_{in}=0.1, M_{out}=0.7)$ 

$$\omega < k_x c (1 - \mathcal{M}_{\mathrm{out}}^2)^{\frac{1}{2}} < k_x c (1 - \mathcal{M}_{\mathrm{in}}^2)^{\frac{1}{2}}$$

in: evanescent out: evanescent

-> decelerated flow  $(M_{in}=0.7, M_{out}=0.1)$ 

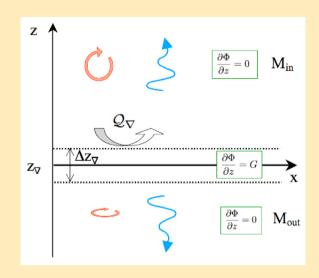
$$k_x c (1 - \mathcal{M}_{\mathrm{in}}^2)^{\frac{1}{2}} < \omega < k_x c (1 - \mathcal{M}_{\mathrm{out}}^2)^{\frac{1}{2}}$$

in: propagate out: evanescent

propagating wave: 
$$\omega > k_x c (1-\mathcal{M}^2)^{rac{1}{2}}$$
 evanescent wave:  $\omega < k_x c (1-\mathcal{M}^2)^{rac{1}{2}}$ 

evanescent wave:

$$\omega^2 = (k_x^2 + k_z^2)c^2 
onumber \ k_z^2 = rac{\omega^2}{c^2} - k_x^2 
onumber \$$



## Compact approximation $\Delta z_{\nabla} \rightarrow 0$

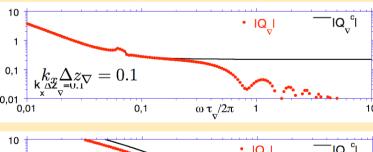
$$\Delta z_
abla o 0$$

scale free

$$au_
abla \equiv \int_{z_
abla - rac{\Delta z_
abla}{2}}^{z_
abla + rac{\Delta z_
abla}{2}} rac{\mathrm{d}z}{|v|}$$

validity of the compact approximation

$$\begin{cases} \tau_{\nabla} \ll \frac{2\pi}{\omega} \\ \\ k_x \Delta z_{\nabla} \ll 1 \end{cases}$$



$$k_x\Delta z_\nabla=1.$$

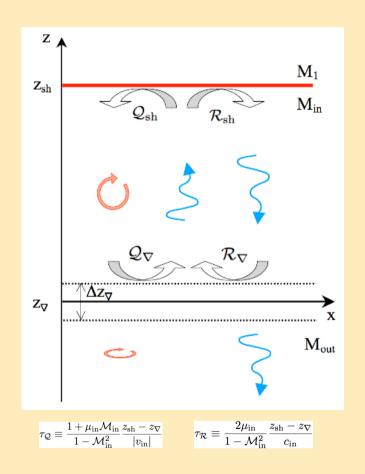
$$\mathcal{Q}_{\nabla}^{\rm c} = \frac{\mathcal{M}_{\rm out} + \mu_{\rm out}}{1 + \mu_{\rm out} \mathcal{M}_{\rm out}} \frac{1}{\mu_{\rm out} \frac{c_{\rm in}^2}{c_{\rm out}^2} + \mu_{\rm in} \frac{\mathcal{M}_{\rm out}}{\mathcal{M}_{\rm in}}} \left[ 1 - \frac{c_{\rm in}^2}{c_{\rm out}^2} + \frac{k_x^2 c_{\rm in}^2}{\omega^2} (\mathcal{M}_{\rm in}^2 - \mathcal{M}_{\rm out}^2) \right]$$

$$\mu^2 \equiv 1 - \frac{k_x^2 c^2}{\omega^2} (1 - \mathcal{M}^2) \qquad \qquad \text{longitudinal} \qquad \text{transverse}$$

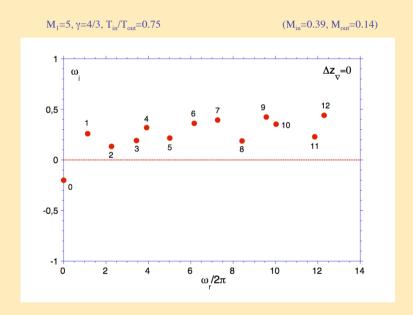
$$\frac{\text{cut-off frequency}}{\sigma_{\nabla}} = \frac{1}{\tau_{\nabla}}$$

-> analytical upper bound of the coupling efficiency

The simplest example of an advective-acoustic instability



-parallel adiabatic flow, localized coupling  $(\gamma, M_1, \Delta\Phi, \Delta z_{\nabla})$ -2-D perturbations  $\omega, k_{\perp}$ 

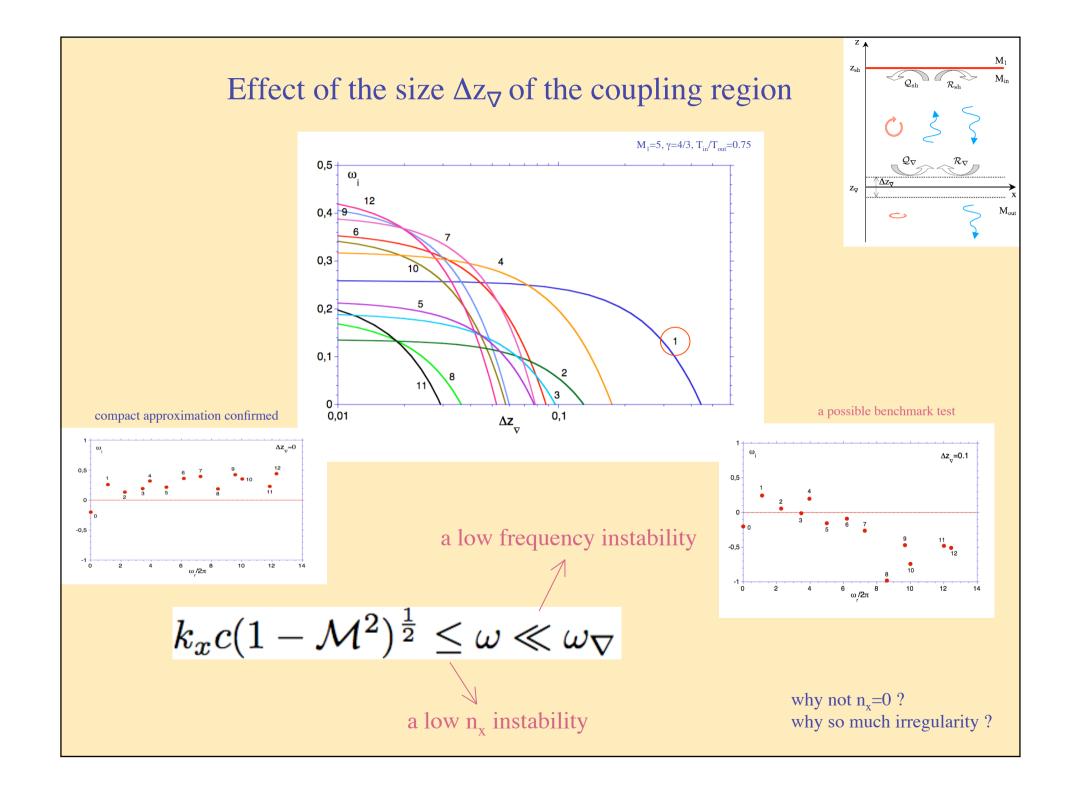


$$\mathcal{Q}e^{i\omega\tau_{\mathcal{Q}}} + \mathcal{R}e^{i\omega\tau_{\mathcal{R}}} = 1$$

$$\mu^2 \equiv 1 - \frac{k_x^2 c^2}{\omega^2} (1 - \mathcal{M}^2)$$

$$R = \frac{\mu_{out} M_{in} c_{in}^{2} - \mu_{in} M_{out} c_{out}^{2}}{\mu_{out} M_{in} c_{in}^{2} + \mu_{in} M_{out} c_{out}^{2}} \left( \frac{1 + \mu_{in} M_{in}}{1 - \mu_{in} M_{in}} \right) \frac{\mu_{in}^{2} - 2\mu_{in} M_{in} + \frac{1}{M_{1}^{2}}}{\mu_{in}^{2} + 2\mu_{in} M_{in} + \frac{1}{M_{1}^{2}}}$$

$$\mathcal{Q}_{c} \equiv \frac{\frac{4\mu_{\rm in}}{\mathcal{M}_{\rm in}}(1-\mathcal{M}_{\rm in}^{2})\left(1-\frac{1}{\mathcal{M}_{1}^{2}}\right)}{\mu_{\rm out}\frac{c_{\rm in}^{2}}{c_{\rm out}^{2}} + \mu_{\rm in}\frac{\mathcal{M}_{\rm out}}{\mathcal{M}_{\rm in}}} \frac{\mathcal{M}_{\rm out} + \mu_{\rm out}}{1+\mu_{\rm out}\mathcal{M}_{\rm out}} \left[ \frac{1-\frac{c_{\rm in}^{2}}{c_{\rm out}^{2}} + \frac{k_{x}^{2}c_{\rm in}^{2}}{\omega^{2}}(\mathcal{M}_{\rm in}^{2} - \mathcal{M}_{\rm out}^{2})}{(\gamma+1)(1-\mu_{\rm in}\mathcal{M}_{\rm in})(\mu_{\rm in}^{2}+2\mu_{\rm in}\mathcal{M}_{\rm in} + \mathcal{M}_{1}^{-2})} \right] \\ R = \frac{\mu_{out}M_{in}c_{in}^{-2} - \mu_{in}M_{out}c_{out}^{-2}}{\mu_{out}M_{in}c_{in}^{-2} + \mu_{in}M_{out}c_{out}^{-2}} \left(\frac{1+\mu_{in}M_{in}}{1-\mu_{in}M_{in}}\right) \frac{\mu_{in}^{-2} - 2\mu_{in}M_{in} + \frac{1}{M_{1}^{2}}}{(\gamma+1)(1-\mu_{in}\mathcal{M}_{in})(\mu_{in}^{2}+2\mu_{in}\mathcal{M}_{in} + \mathcal{M}_{1}^{-2})} \right]$$



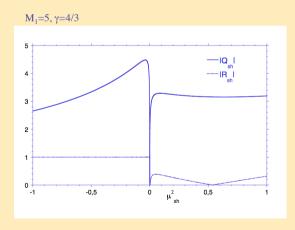
## Comparison of $n_x=0$ and $n_x>0$ modes

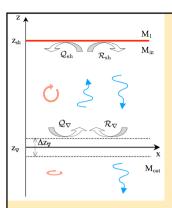
- no vorticity in 1-D: the instability of the mode  $n_x=0$  relies only on temperature gradients
- transverse modes benefits from the vortical-acoustic coupling

$$\mathcal{Q}_{\mathrm{c}} \equiv \frac{\frac{4\mu_{\mathrm{in}}}{\mathcal{M}_{\mathrm{in}}}(1-\mathcal{M}_{\mathrm{in}}^2)\left(1-\frac{1}{\mathcal{M}_{\mathrm{1}}^2}\right)}{\mu_{\mathrm{out}}\frac{c_{\mathrm{in}}^2}{c_{\mathrm{out}}^2} + \mu_{\mathrm{in}}\frac{\mathcal{M}_{\mathrm{out}}}{\mathcal{M}_{\mathrm{in}}}} \frac{\mathcal{M}_{\mathrm{out}} + \mu_{\mathrm{out}}}{1+\mu_{\mathrm{out}}\mathcal{M}_{\mathrm{out}}} \left[\frac{1-\frac{c_{\mathrm{in}}^2}{c_{\mathrm{out}}^2} + \frac{k_x^2c_{\mathrm{in}}^2}{\omega^2}(\mathcal{M}_{\mathrm{in}}^2-\mathcal{M}_{\mathrm{out}}^2)}{(\gamma+1)(1-\mu_{\mathrm{in}}\mathcal{M}_{\mathrm{in}})(\mu_{\mathrm{in}}^2+2\mu_{\mathrm{in}}\mathcal{M}_{\mathrm{in}}+\mathcal{M}_{\mathrm{1}}^{-2})}\right]$$

At the shock, the coupling efficiency is also maximum for transverse perutrbations:

 $\mu_{
m sh}^2 \propto \pm rac{1}{\mathcal{M}_1^2}$ 





### Contribution of the acoustic cycle

(Foglizzo 2002)

$$\mathcal{Q}e^{i\omega\tau_{\mathcal{Q}}} + \mathcal{R}e^{i\omega\tau_{\mathcal{R}}} = 1$$

global dispersion relation

The acoustic cycle alone is stable:  $|\mathcal{R}| \leq 1$ 

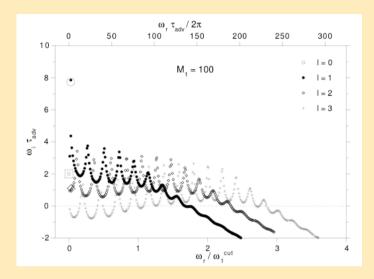
Its contribution can be either constructive or destructive

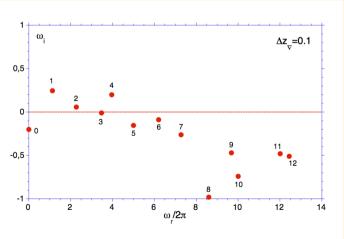
$$|\mathcal{Q}| + |\mathcal{R}| < 1$$
 stable

$$|\mathcal{Q}| - |\mathcal{R}| > 1$$
 unstable

$$lpha \equiv rac{|\mathcal{R}|}{|\mathcal{Q}|^{rac{ au_{\mathcal{R}}}{ au_{\mathcal{Q}}}}}$$

$$\frac{1}{\tau_{\mathcal{Q}}}\log\left\{(1-\alpha)|\mathcal{Q}|\right\} \le \omega_i \le \frac{1}{\tau_{\mathcal{Q}}}\log\left\{\frac{|\mathcal{Q}|}{1-\alpha}\right\}$$





->the stability threshold is very sensitive to geometrical factors

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#### Conclusion

Simple toy model of the advective-acoustic instability in a decelerated flow

- gradient cut-off -> low frequency

$$k_x c (1 - \mathcal{M}^2)^{\frac{1}{2}} \le \omega \ll \omega_{\nabla}$$

- acoustic evanescence ->  $low n_x$
- vorticity -> transverse rather than longitudinal
- acoustic cycle -> sensitive to geometrical parameters

Benchmark test for numerical simulations

- advection of vorticity, numerical viscosity?

Relevance to the core-collapse problem?

- gravity, geometry, photodissociation, heating, cooling
- detailed comparison with numerical simulations: ongoing effort (Blondin, Scheck et al. 2006)

